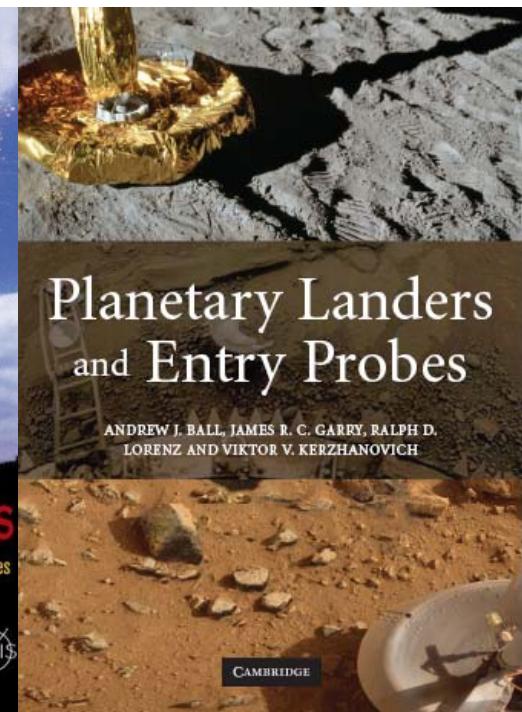
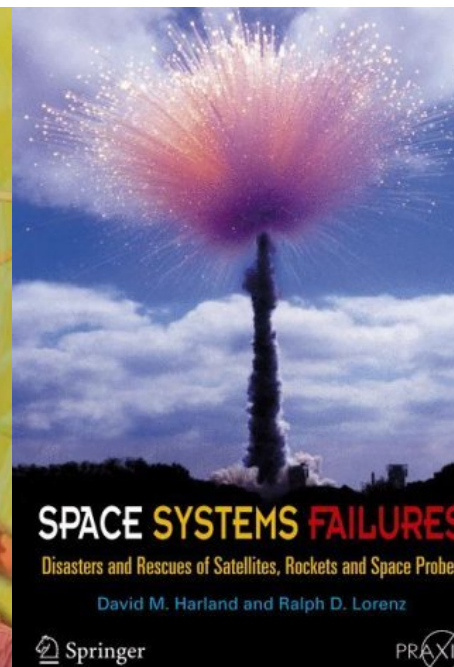
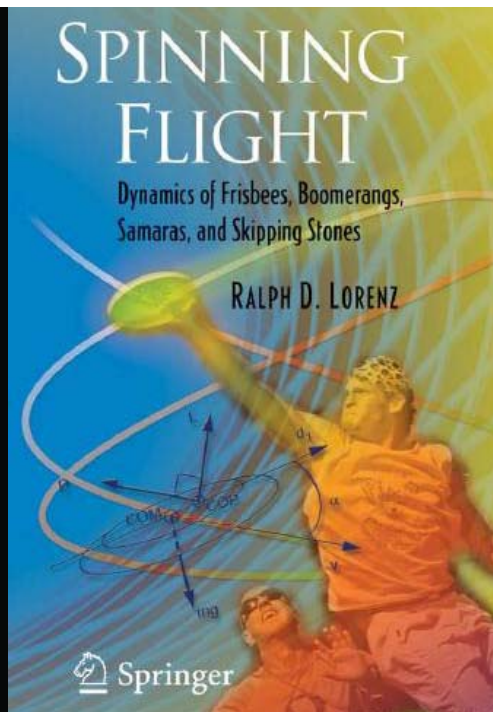
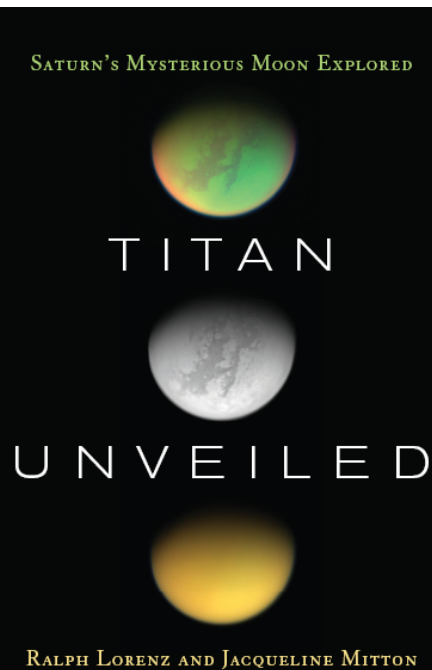


Acoustics Measurements on the Huygens Probe and other Platforms - A Review

Ralph Lorenz
JHU-APL



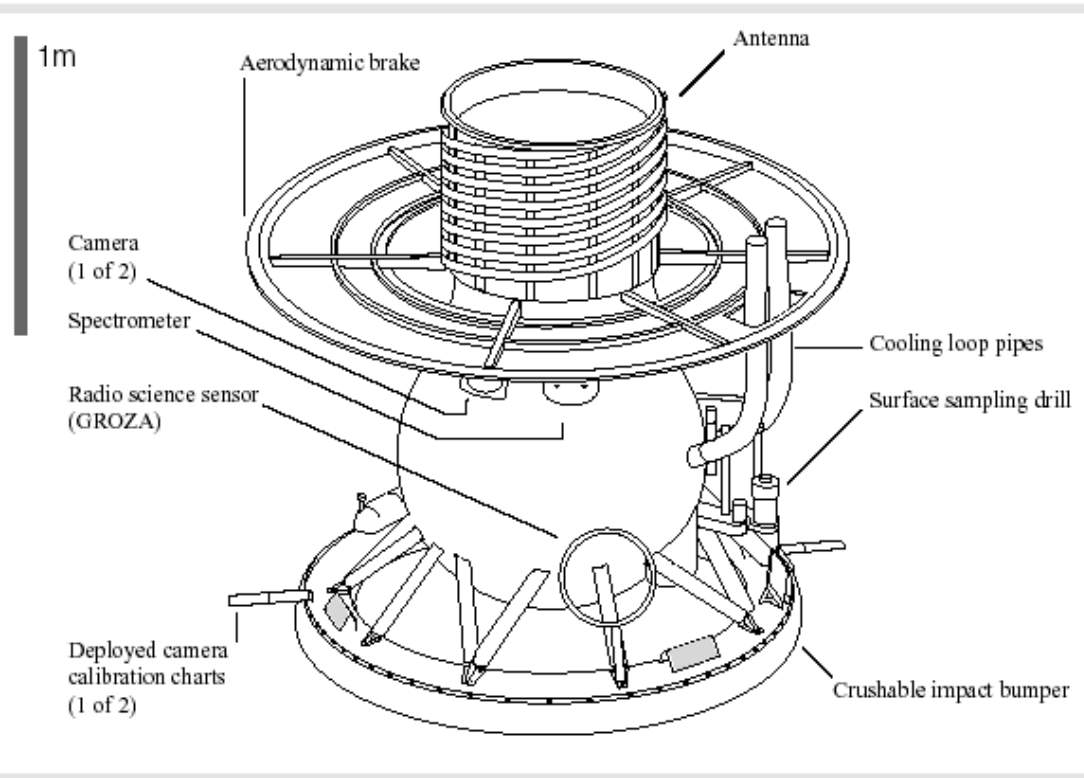


Figure 18.9 Venera 11, 12.

Figure from Ball et al., Planetary Landers and Entry Probes, CUP, 2007

'Groza' instrument on
Venera 11,12.
(Groza-2 on V13/14)
Passive acoustic
sensing to detect
thunder.

The Groza 2 experiment aboard the landing craft comprised a microphone and appropriate electronic circuitry for measuring acoustic noise both along the descent path and on the surface of the planet. The electromagnet-type microphone remained operational at temperatures up to 800 K and pressures up to 100 bar. The microphone accommodated a 2-kHz frequency band, from 400 to 2500 Hz, with peak sensitivity near 1700 Hz (under standard conditions).

Descent signals attributed to aeroacoustic noise.

Acoustic measurements of the wind velocity at the *Venera 13* and *Venera 14* landing sites

L. V. Ksanfomaliti, N. V. Goroshkova, M. K. Naraeva, A. P. Suvorov, V. K. Khondyrev, and L. V. Yabrova

Institute for Space Research, USSR Academy of Sciences, Moscow

(Submitted May 17, 1982)

Pis'ma Astron. Zh. 8, 419-423 (July 1982)

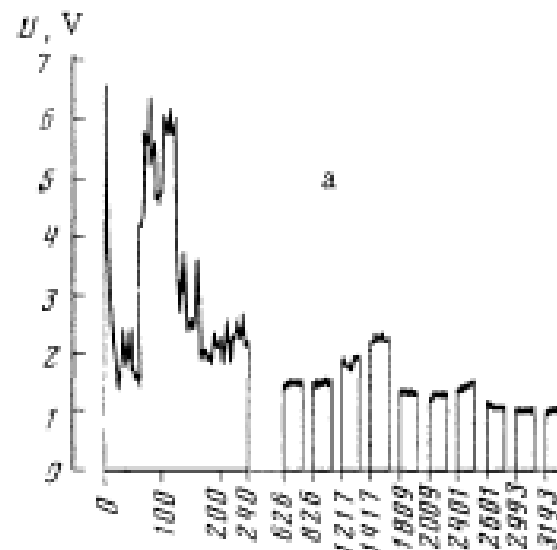
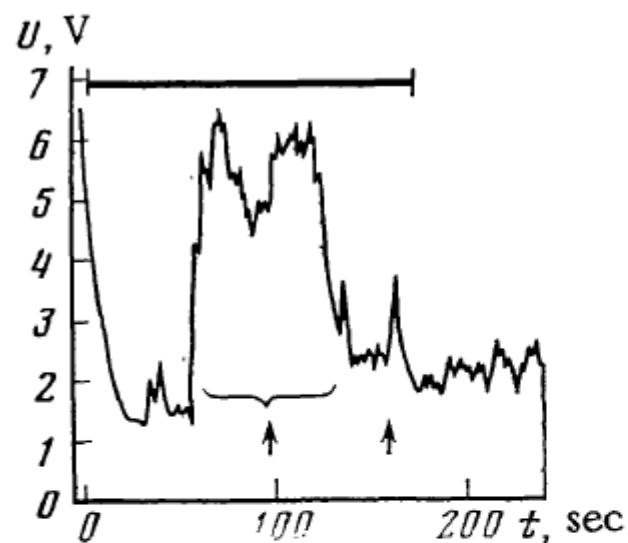
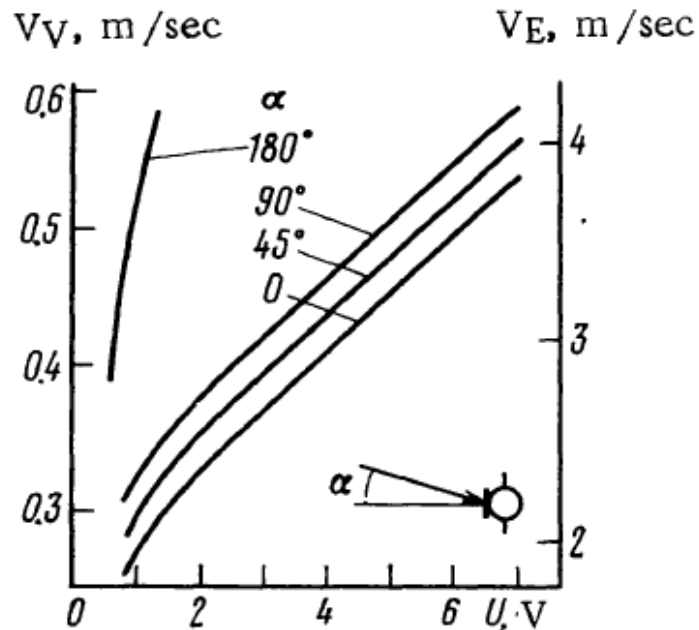


FIG. 1. Acoustic noise recorded by the *Venera 14* lander on the surface of Venus, March 5, 1982. Vertical axis, output signal U (the full scale corresponds to the range from 55 to 82 dB; horizontal axis, elapsed time after landing. The heavy bar indicates the period when the spacecraft systems were operating. The arrows indicate noise associated with equipment operation. Presumably the signal from $t = 180$ to 240 sec represents wind noise in the microphone armature.

By interpreting the acoustic noise on the Venus surface as wind noise in the microphone armature, estimated wind velocities of 0.35-0.57 m/sec are obtained, in agreement with earlier measurements on other spacecraft. It is noted that these values are also consistent with the observed drift of fine soil particles across the surface of the spacecraft landing ring.



Dynamic Pressure
scaling ρV^2

FIG. 2. Output signal U as a function of the wind velocity V_V on Venus and V_E on the earth when the microphone is employed as an anemometer. The angle α specifies the direction of the wind relative to the microphone diaphragm.

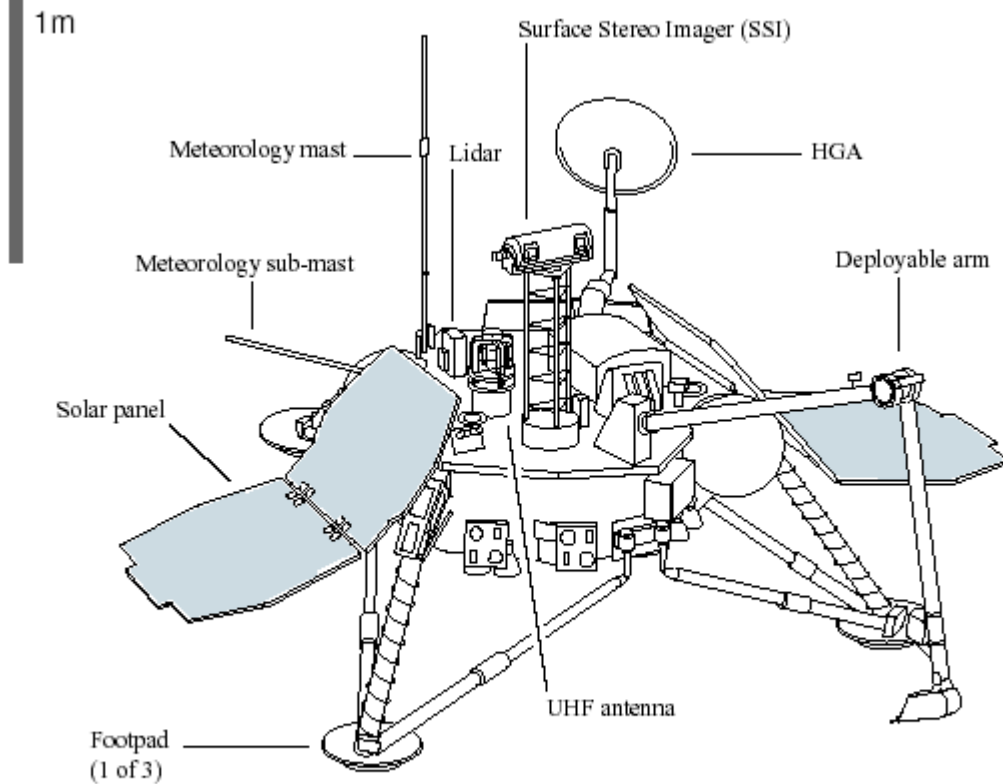
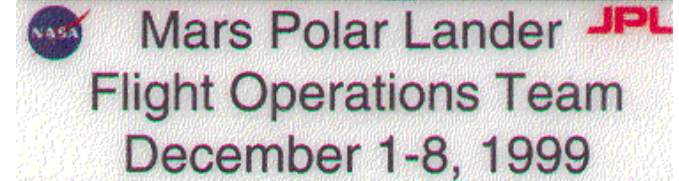
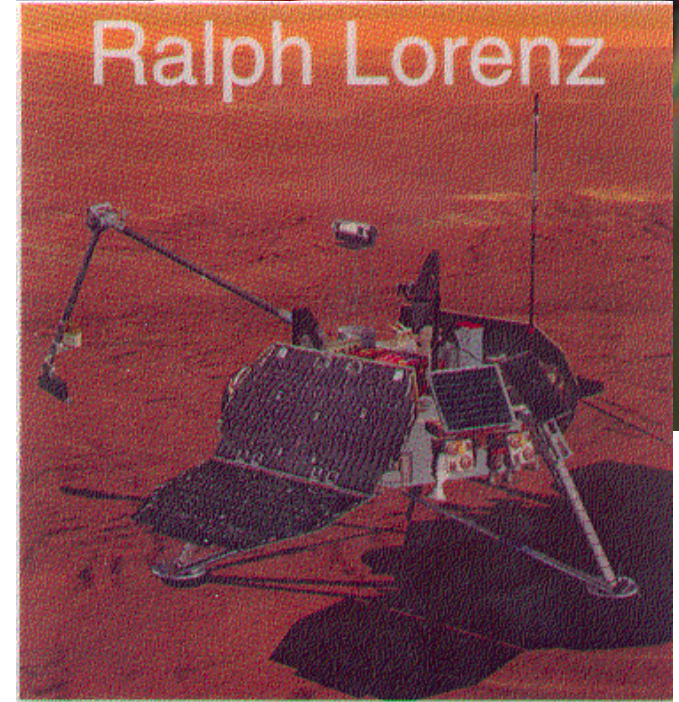
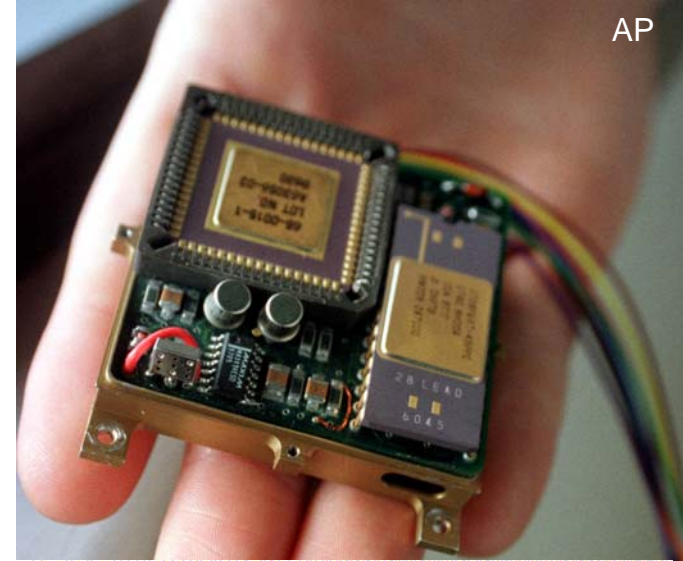


Figure 18.13 Mars Polar Lander.

Figure from Ball et al., Planetary Landers and Entry Probes, CUP, 2007

UC Berkeley/The Planetary Society
 Mars Microphone on Mars Polar Lander
 (piggybacked on LIDAR) 0.050kg (<\$100K?)
 Variant developed for Netlander



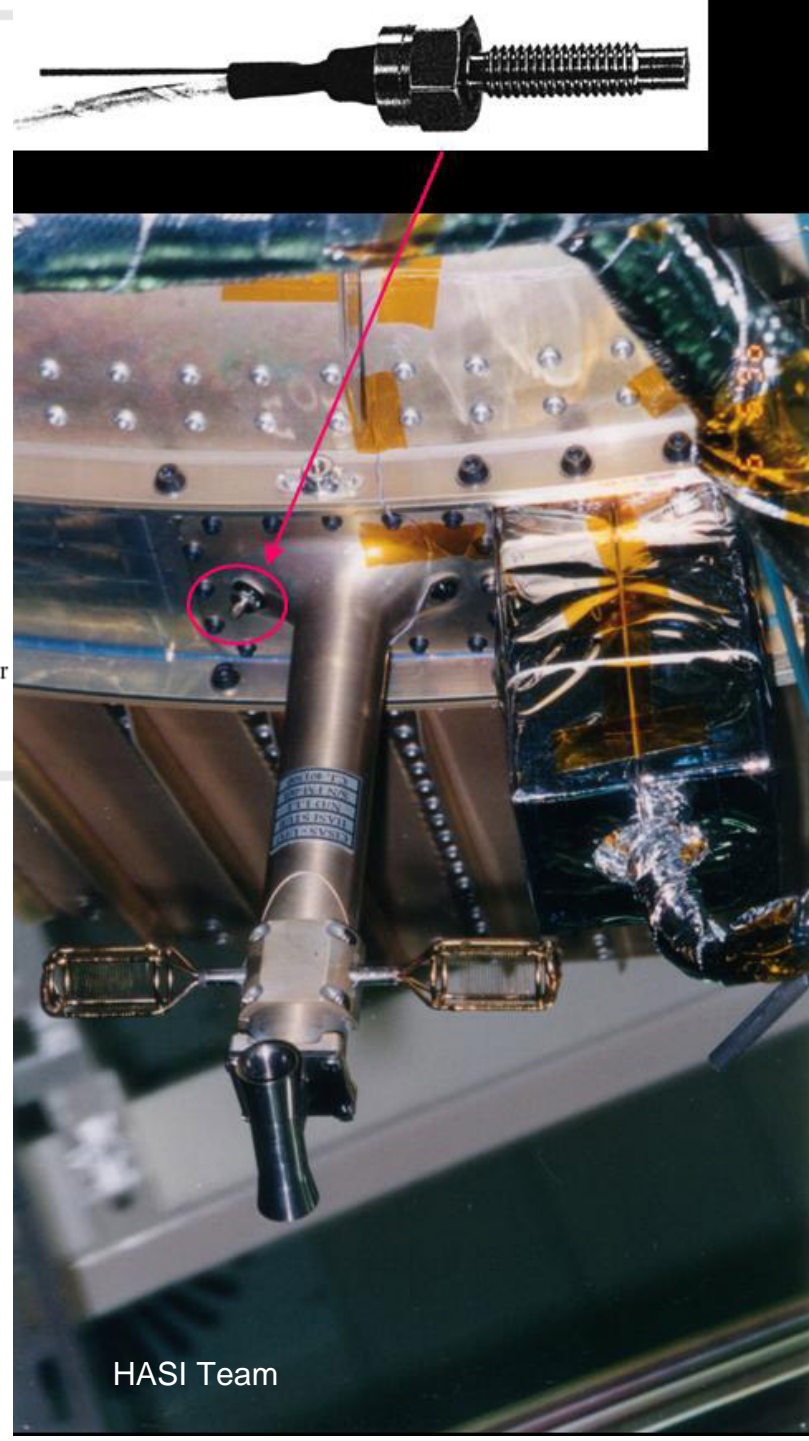
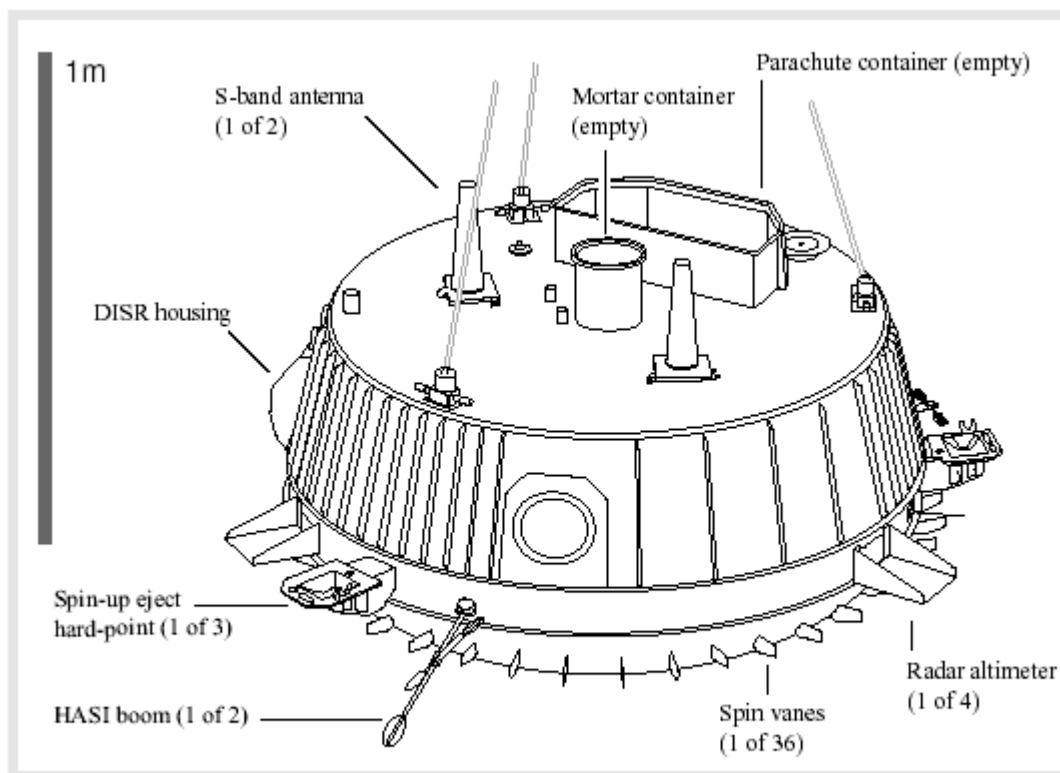


Figure 16.10 Huygens. Figure from Ball et al., Planetary Landers and Entry Probes, CUP, 2007

Microphone processed by DSP in HASI Permittivity/Wave Analyzer

Overage 480 bps (cf 128 kbps MP3...)

HASI Team

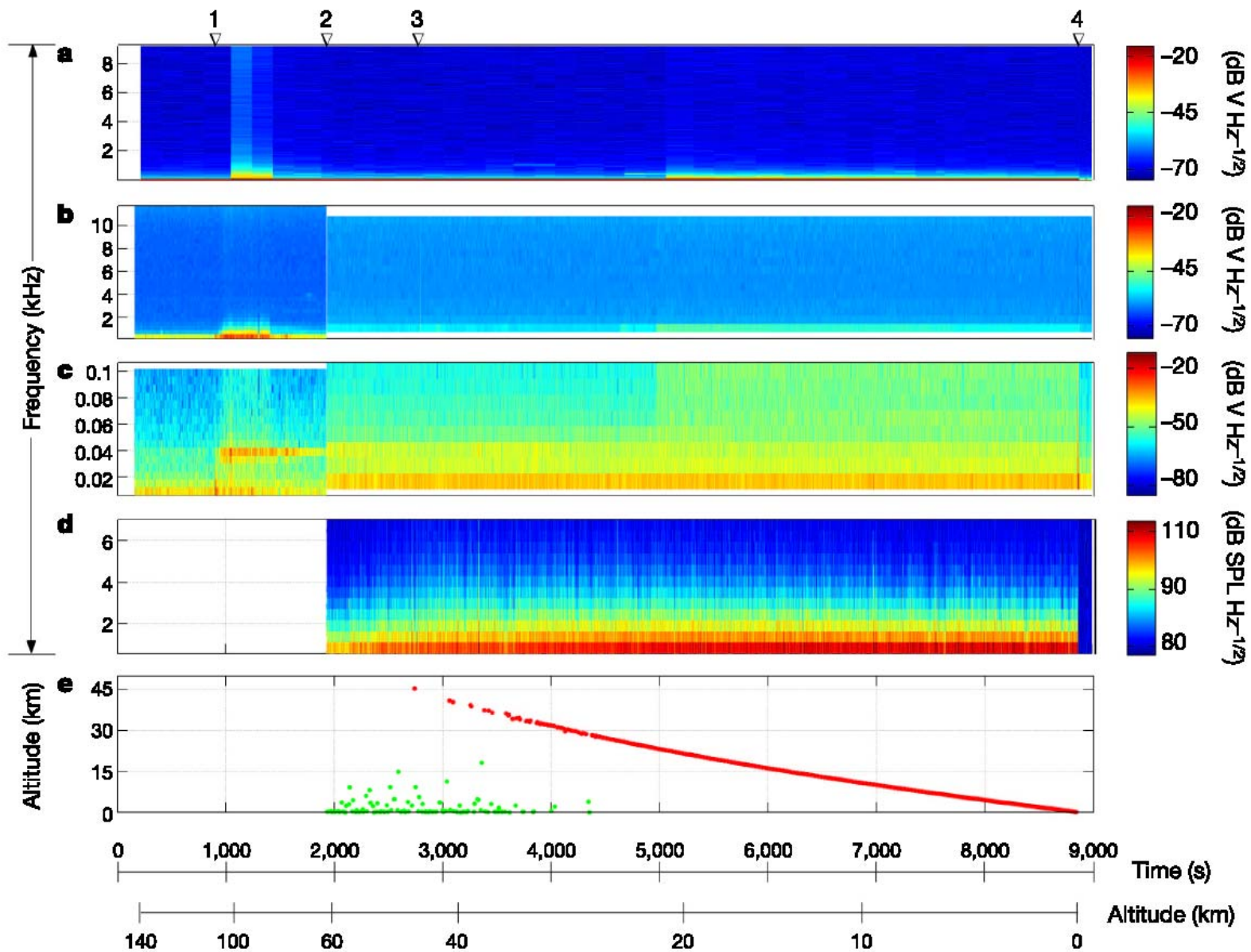
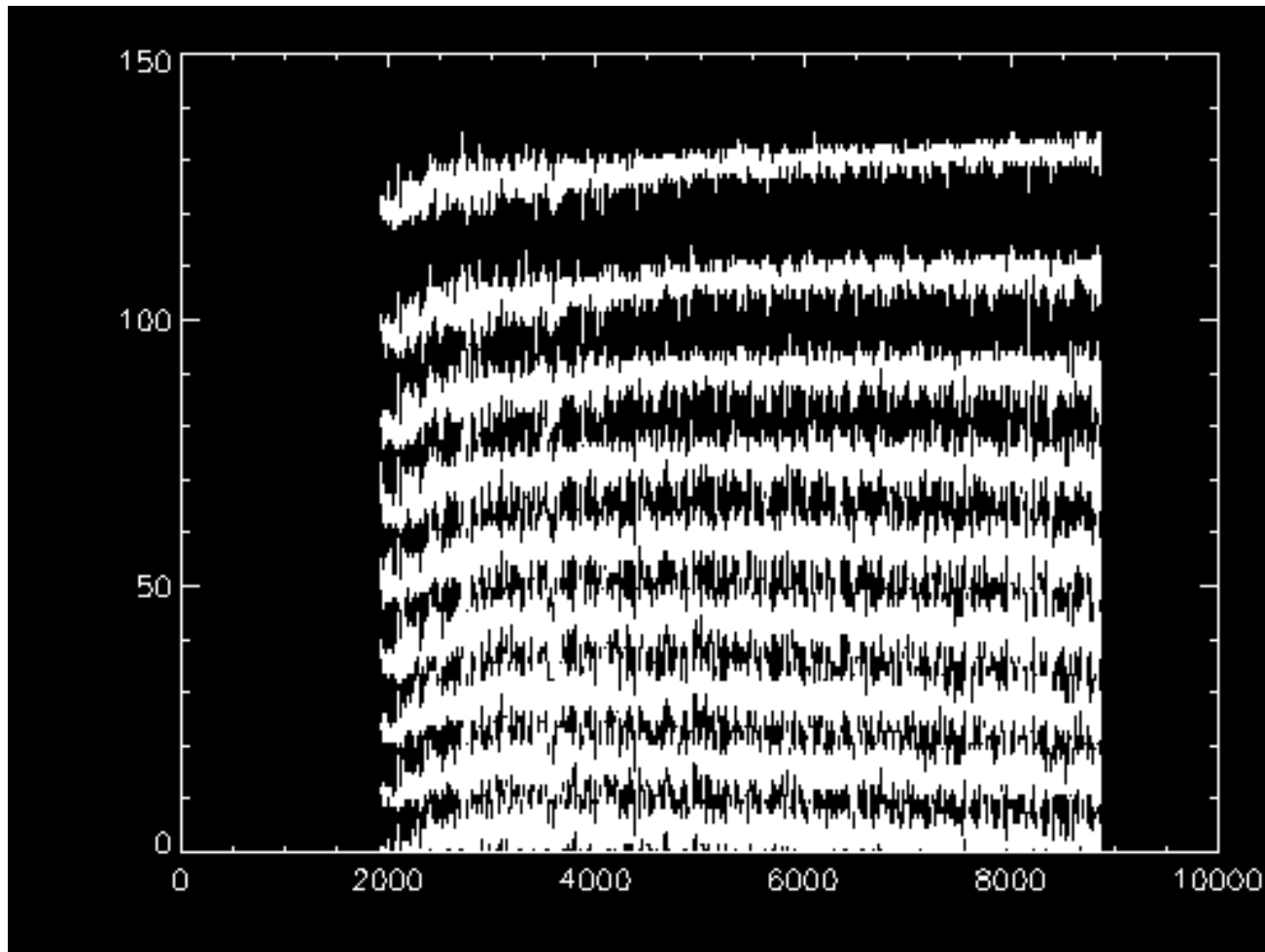


Figure from Fulchignoni et al., Nature, 2005

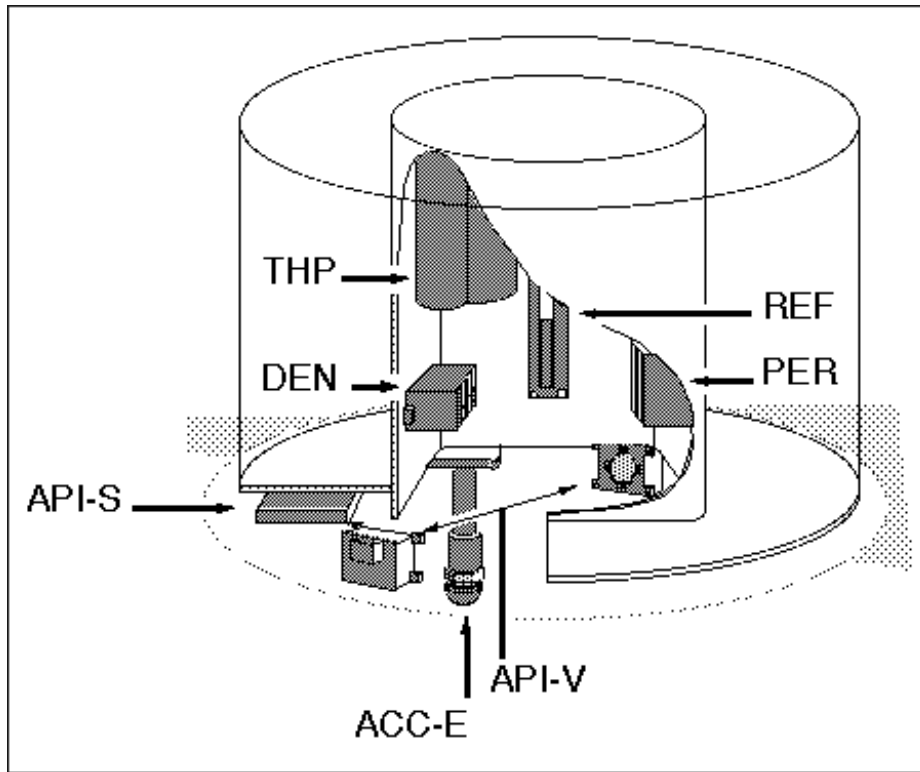
Signal more or less constant during descent. Why ?

Sound level corresponds to fluctuations in pressure sensed at microphone - forced by dynamic pressure during descent. Under a given parachute in steady descent dynamic pressure is simply weight per unit area - roughly constant (g , Re , M etc....)

Possible acoustic impedance effect at beginning ?



SSP 'TopHat' Layout

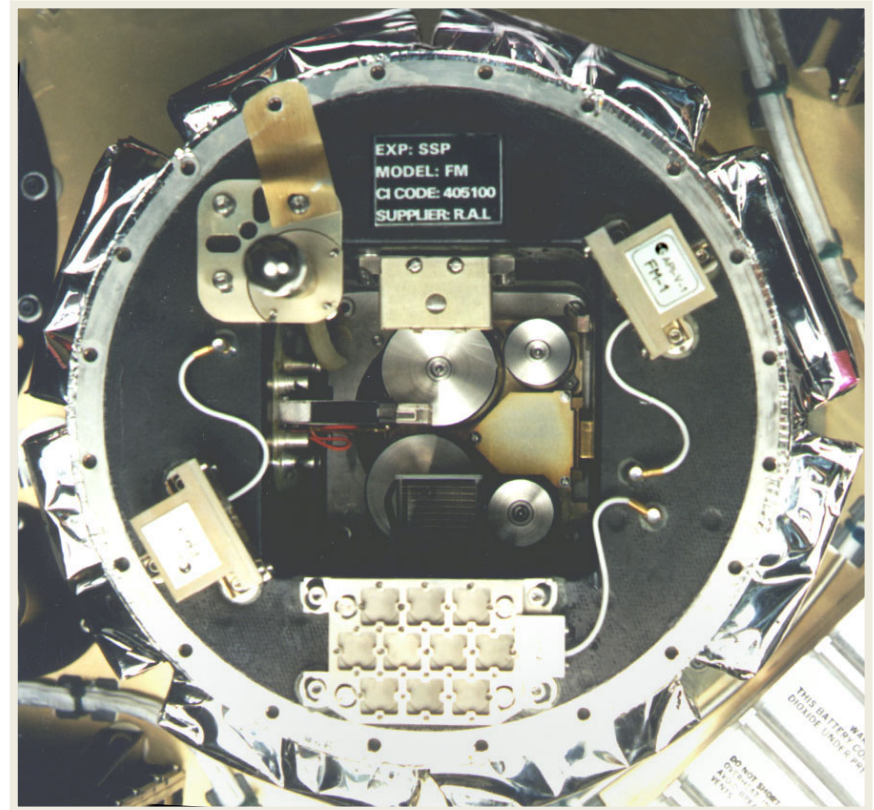


API-V: speed of sound in media

API-S: downward-looking sonar

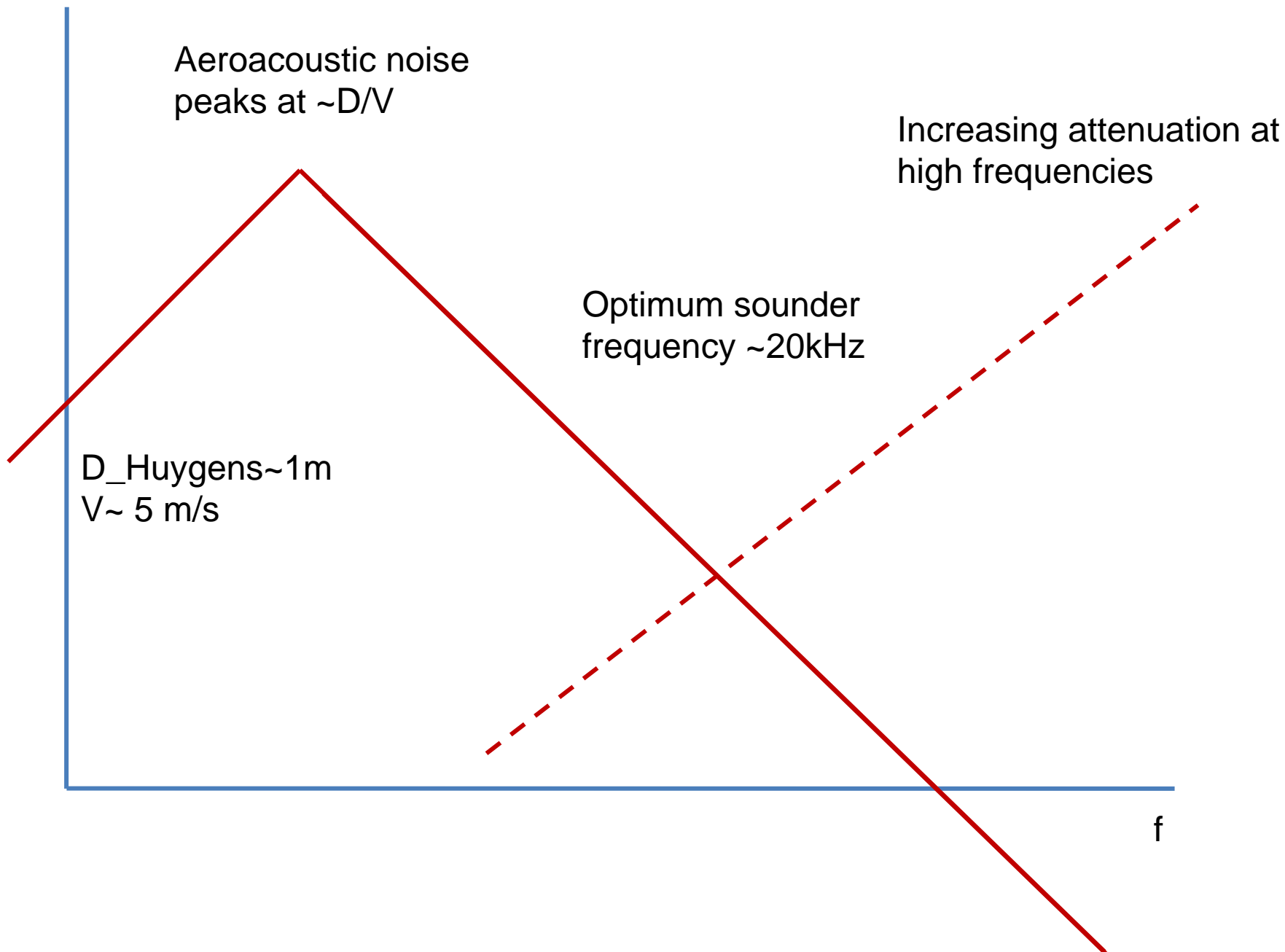
THP: thermal properties of media

PER: electrical permittivity /
conductivity
REF: refractive index of
liquid



DEN: density of liquid

ACC-E: landing impact force sensor
(ACC-I: acceleration on impact
TIL: angle of probe) - on electronics
box



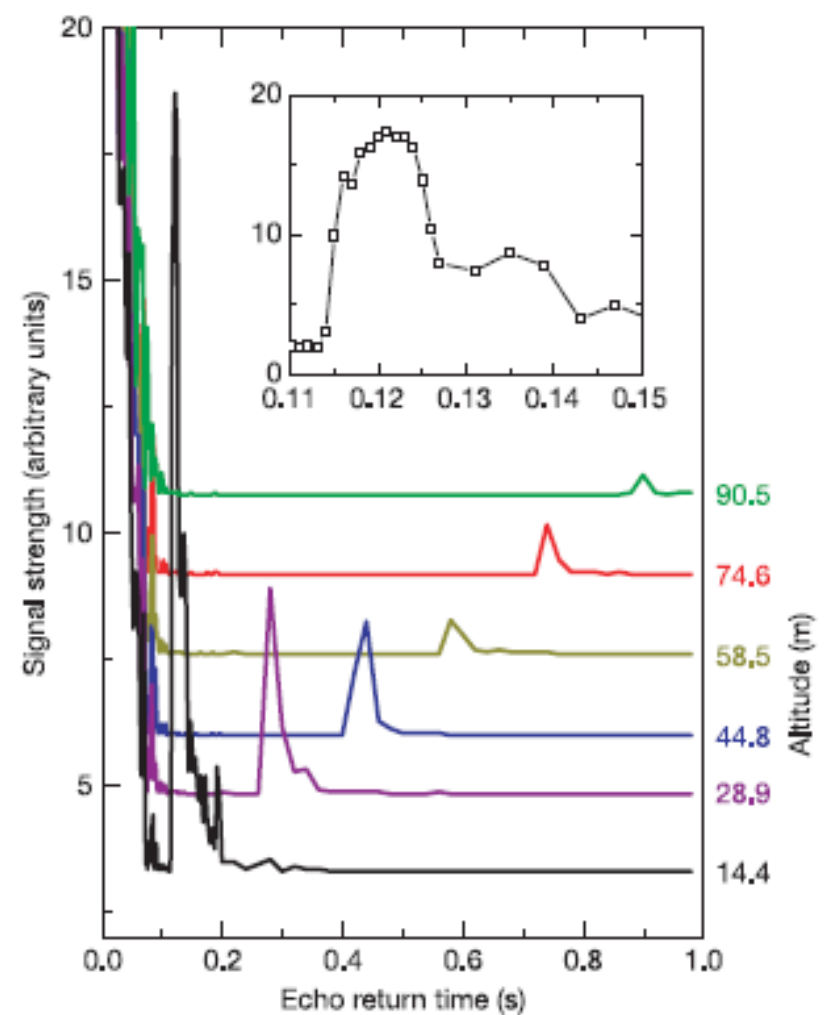
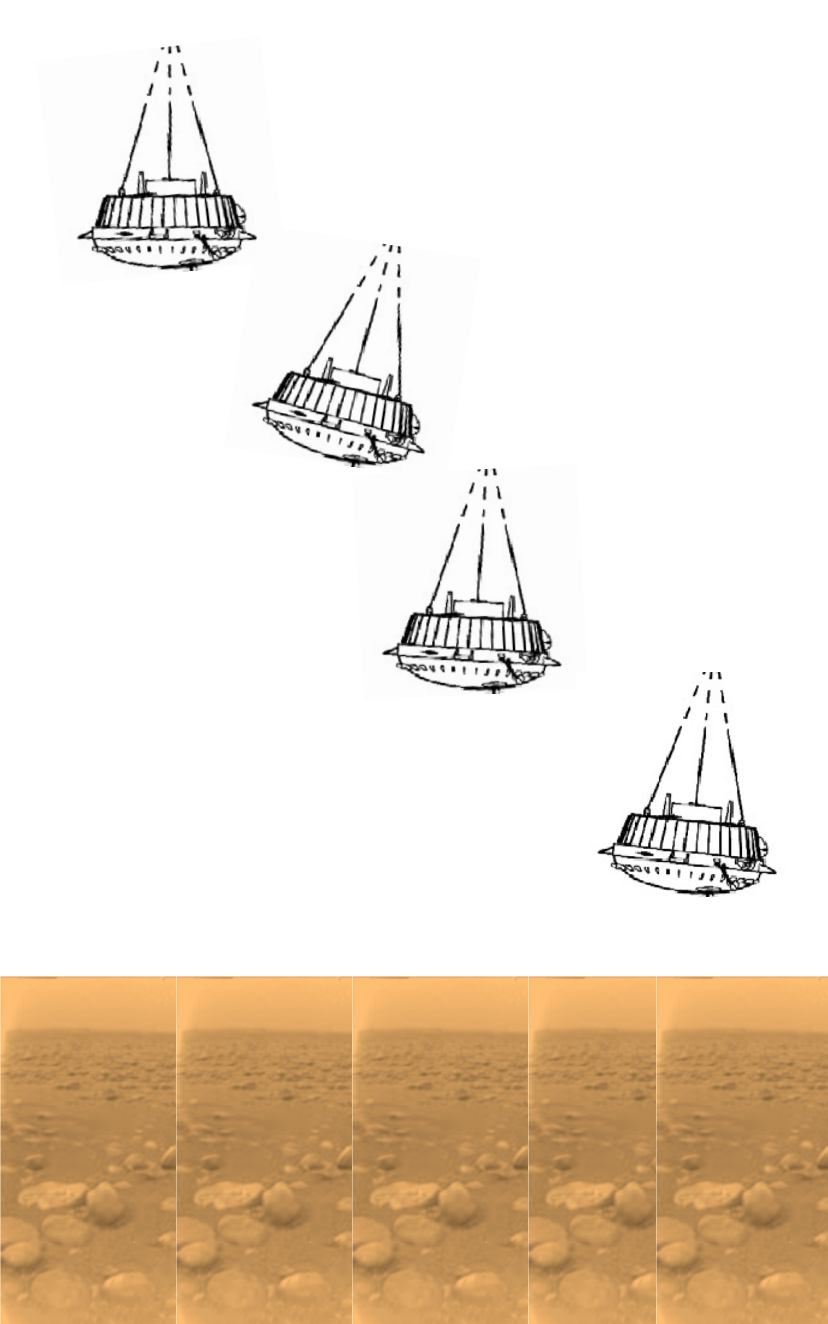
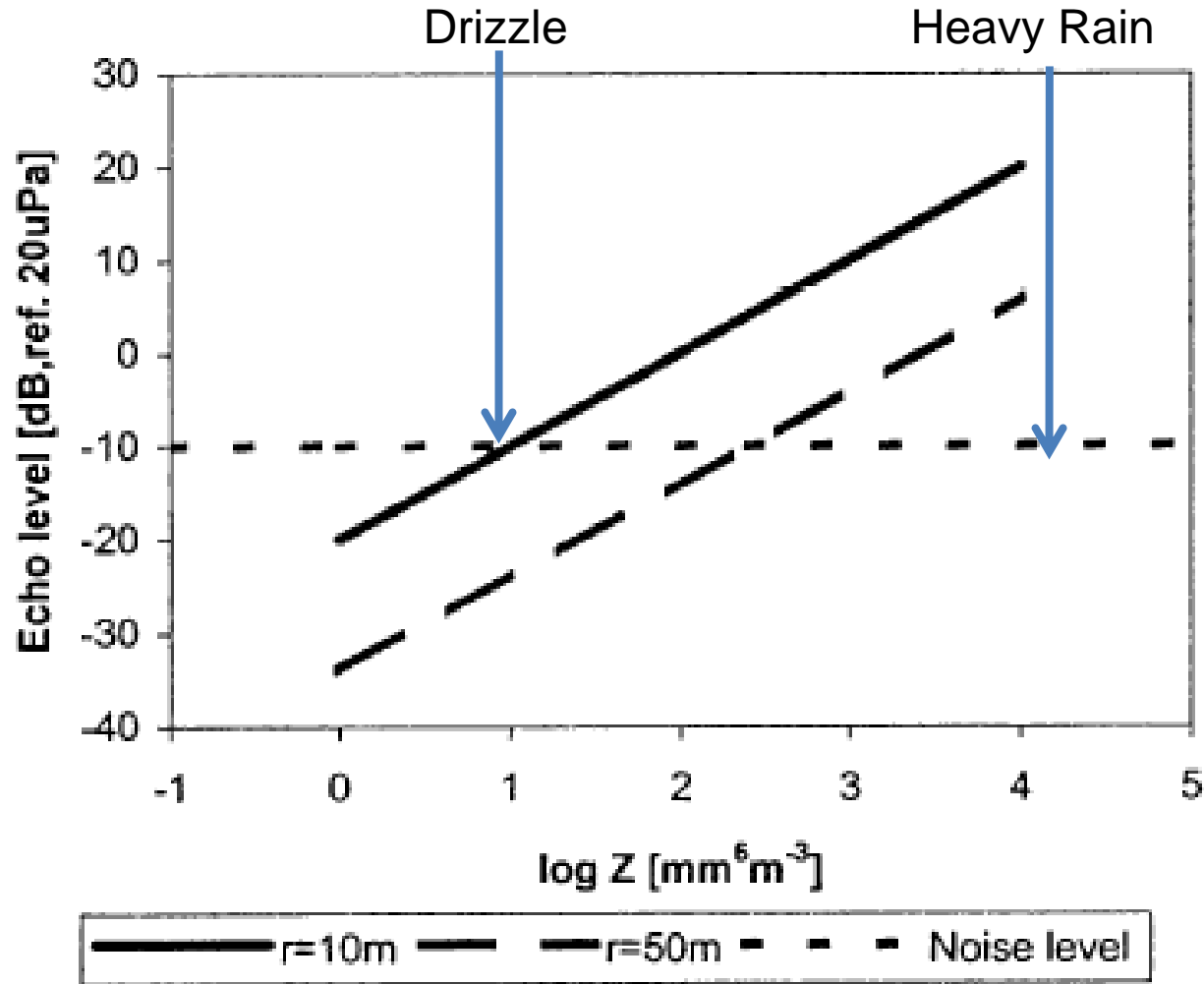


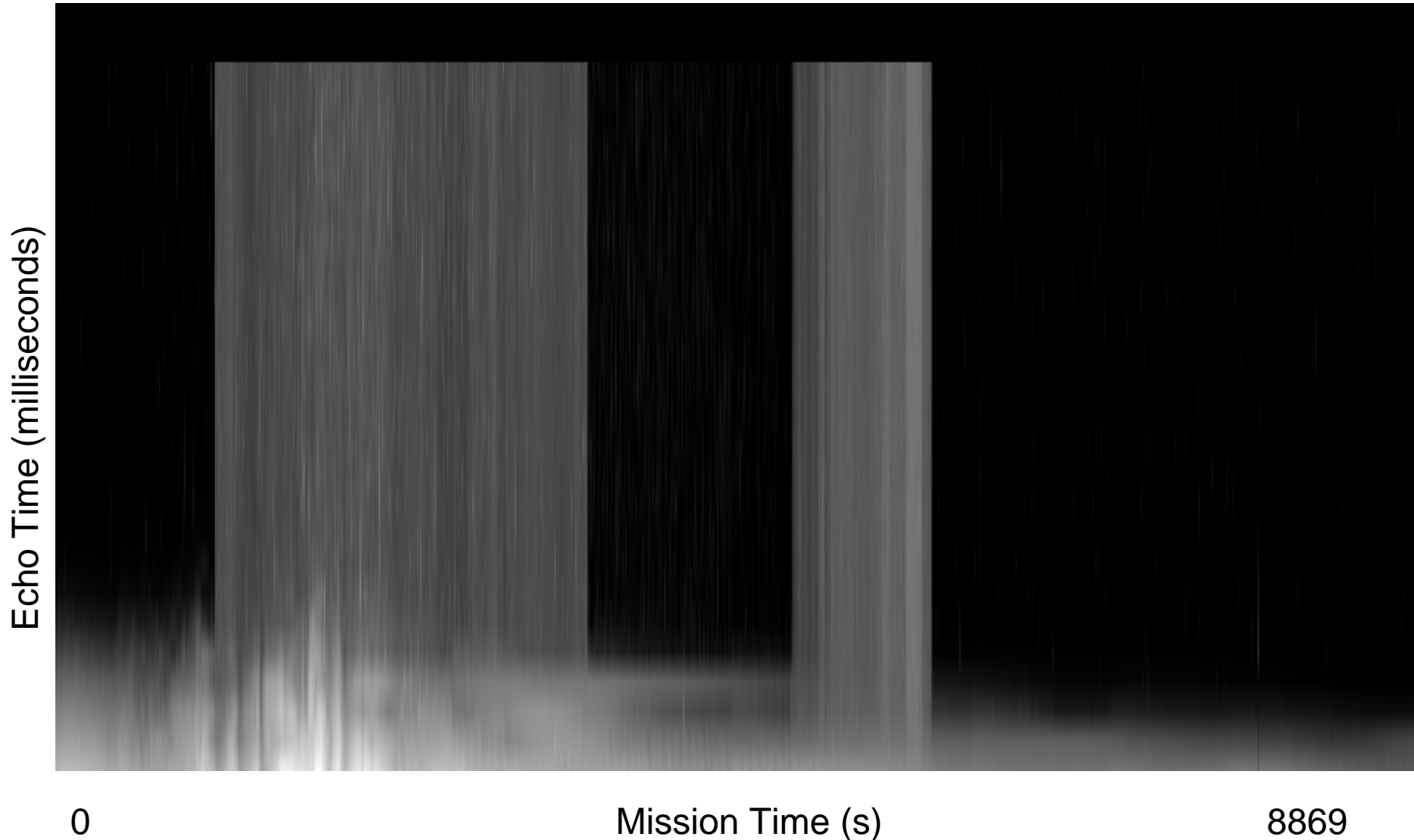
Figure 1 | Acoustic sonar (API-S) surface echoes. Note that the larger signals to the left of the plot are the result of the sensor ringing from the send pulse intruding into the receive time window. The inset is a zoom on the final API-S surface detection from 14.4 m altitude (at the time of pulse transmission). A speed of sound measurement of $191.9 \pm 1.8 \text{ m s}^{-1}$ from the SSP Acoustic Velocity (API-V) sensors near the surface is used to convert ranging time delay into altitude.

SSP API-S as Weather 'Radar'

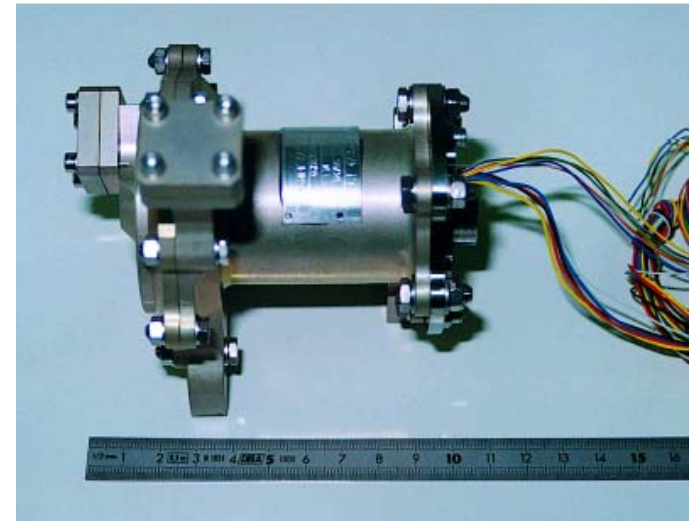
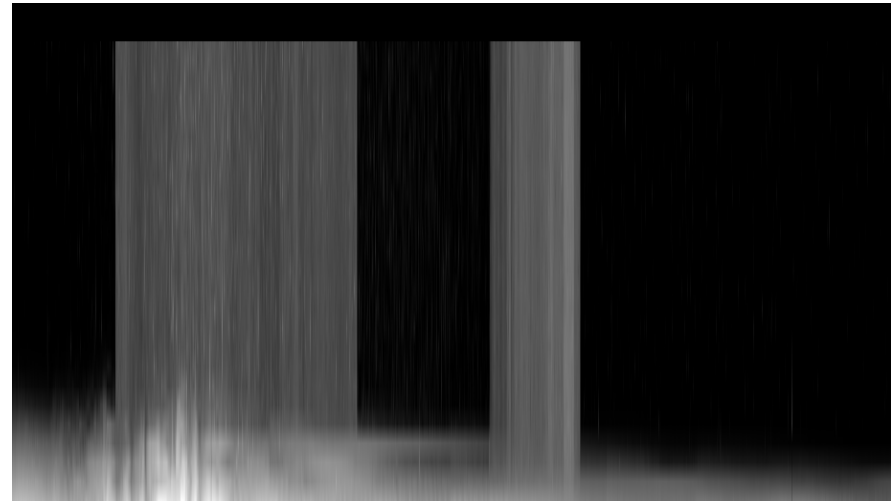
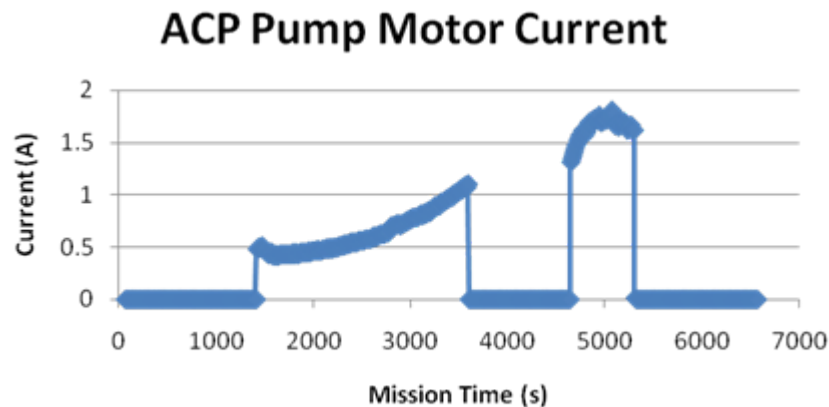
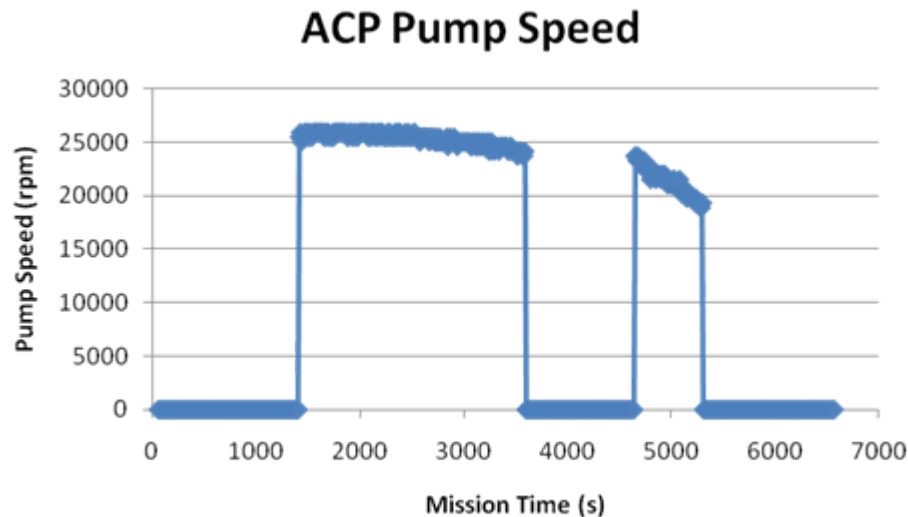


From Svedhem et al., Planetary Probe Atmospheric Entry and Descent Trajectory Analysis and Science, Lisbon, PT 2002 (aka IPPW-1) Proceedings ESA SP-544

Search for atmospheric backscatter (turbulence, precipitation). 'Jammed' by external noise during two major episodes of descent

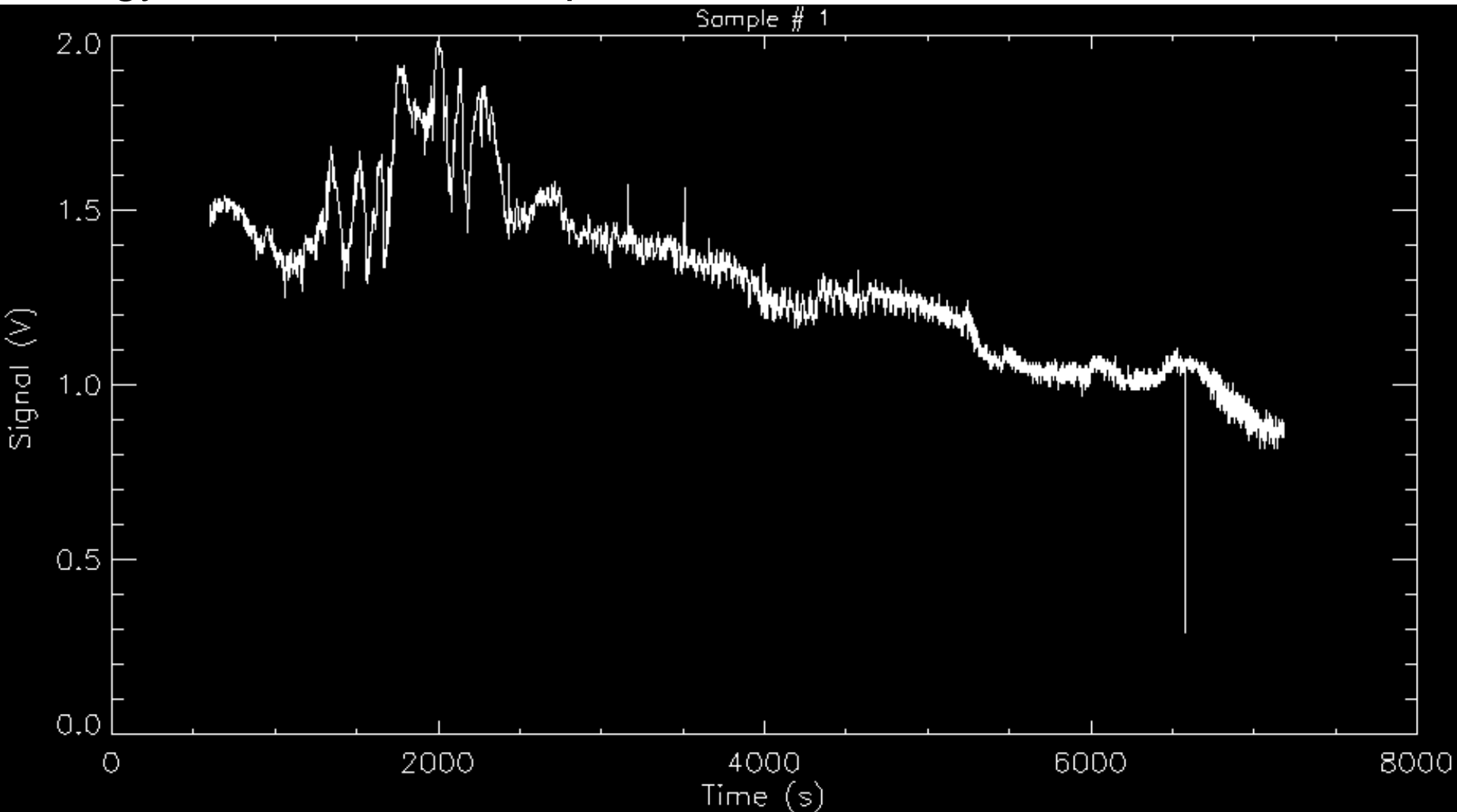


Descent mode observations dominated by noise from ACP pump

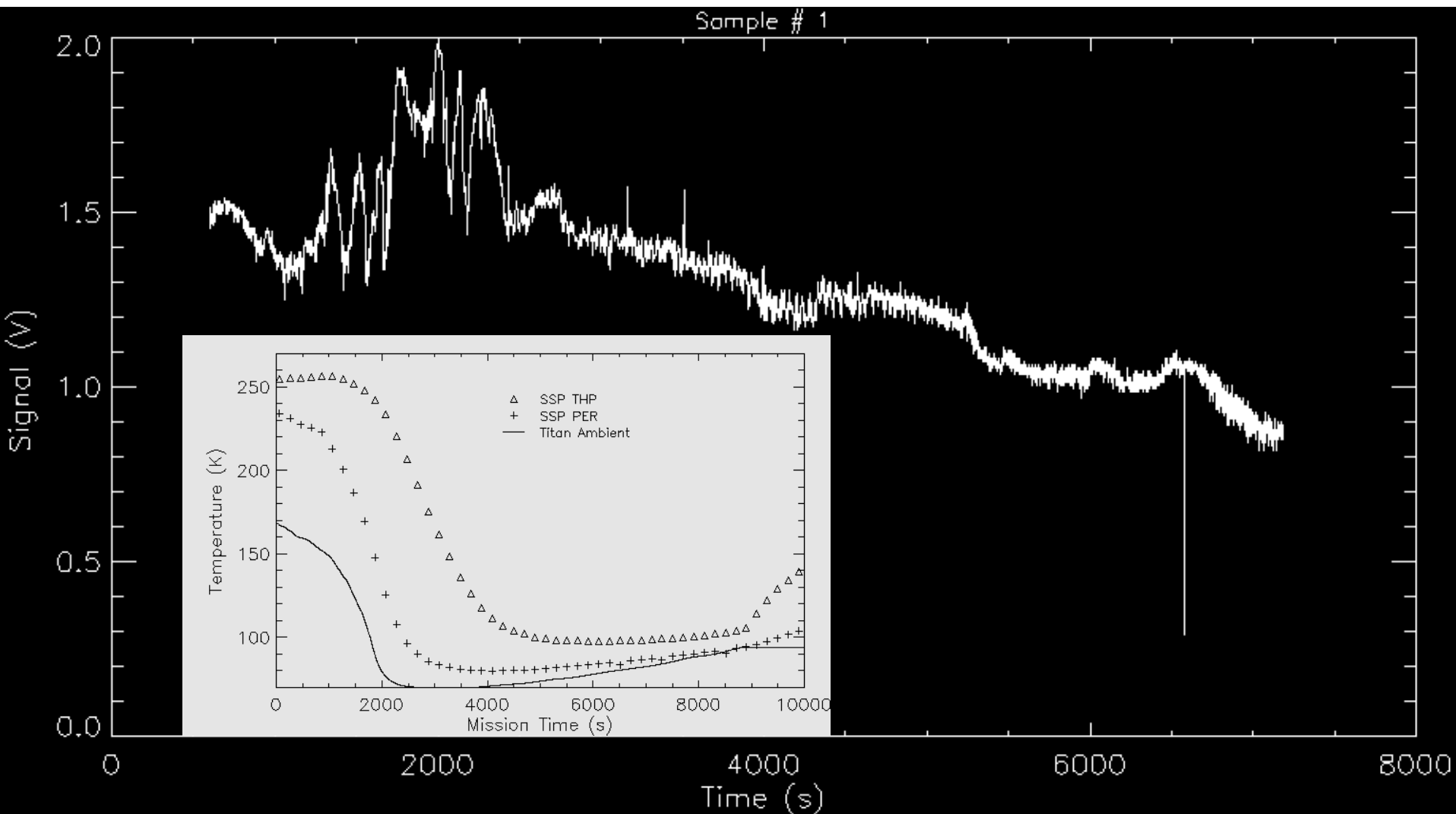


Tx ringing was noted to increase after launch: transducer elements shifted ?

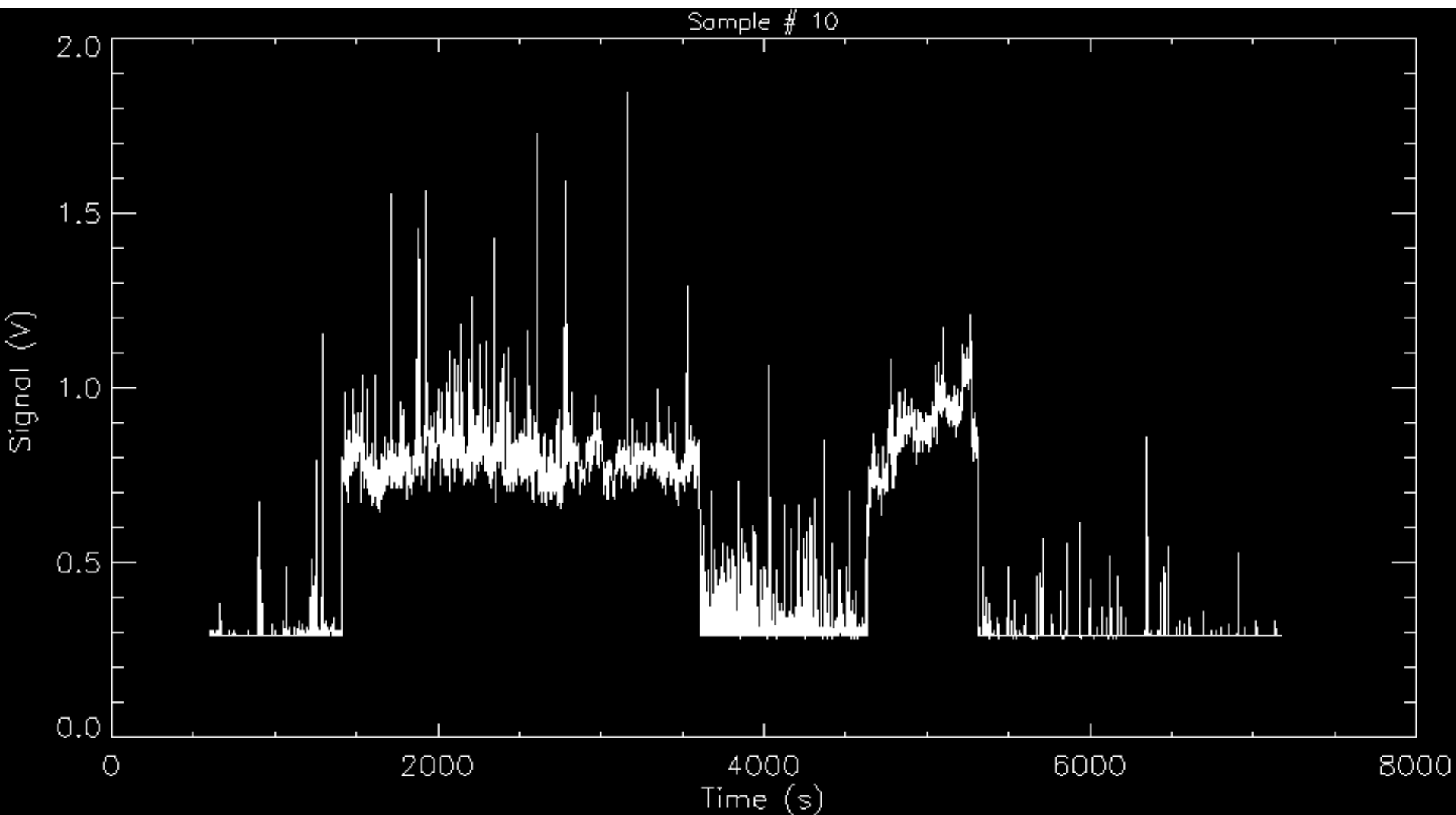
Signal in first bin (ringing) declines with time - better coupling of energy into denser atmosphere?

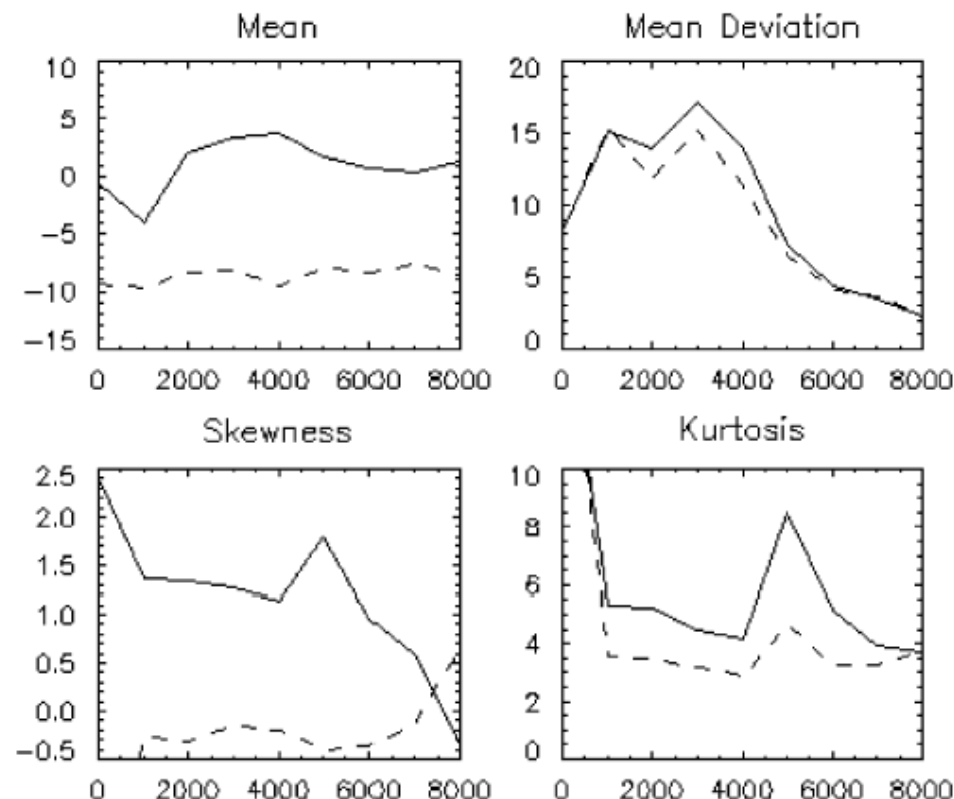
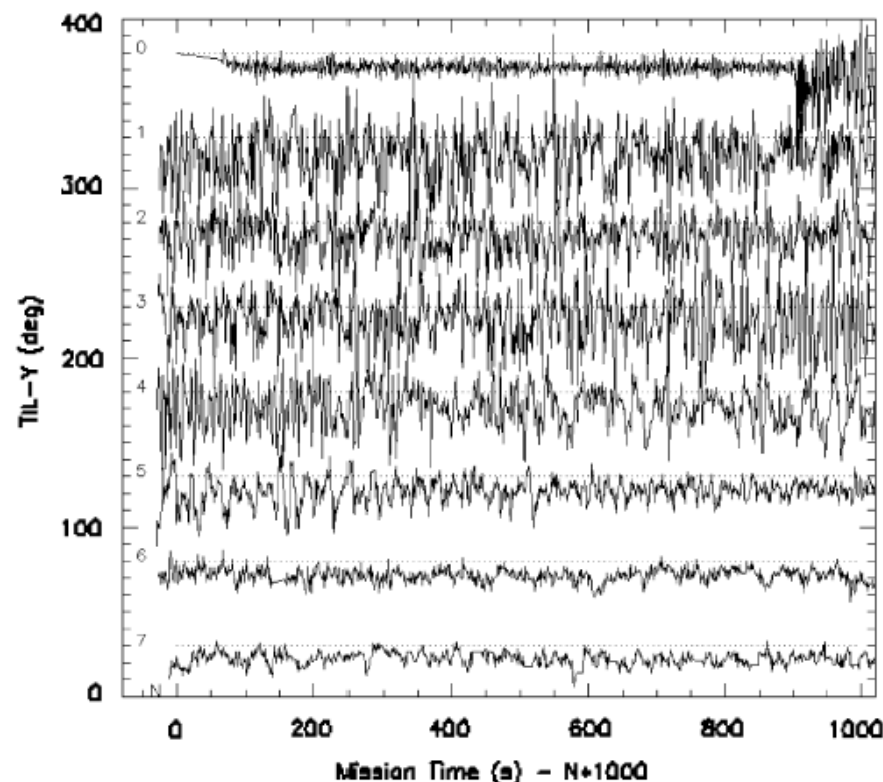


Odd character 1000-2500s..... Period of most rapid external temperature change... (although other things going on too..)



'base level' of signal in last bin dominated by ACP noise, but non-monotonic trend in the three 'quiet' regions - 3700-4500s region has much more echo power than early and late phases





Available online at www.sciencedirect.com



Planetary and Space Science ■■■■■■■■■■

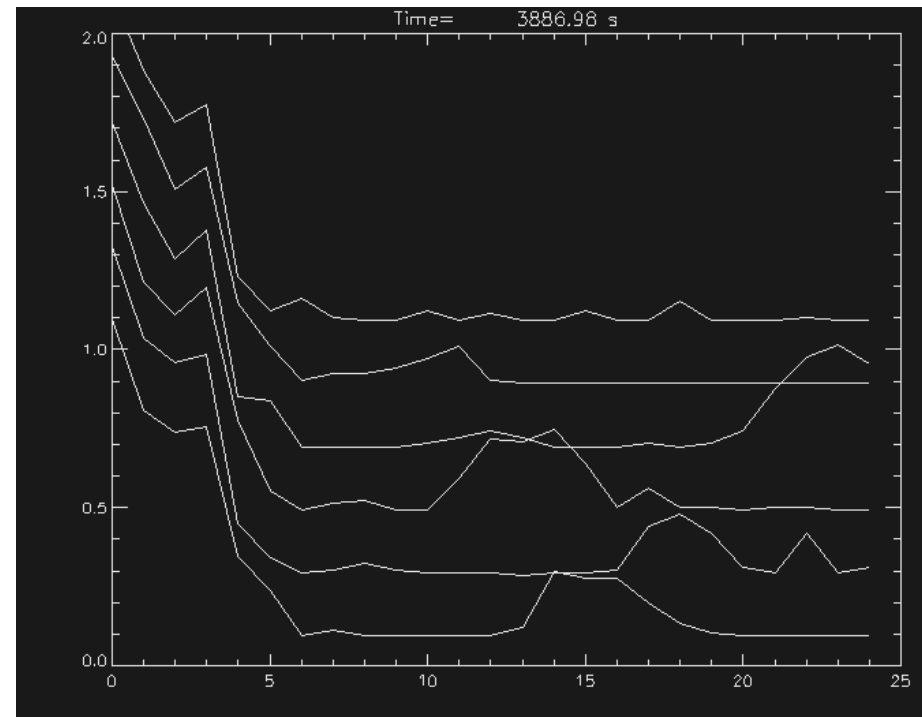
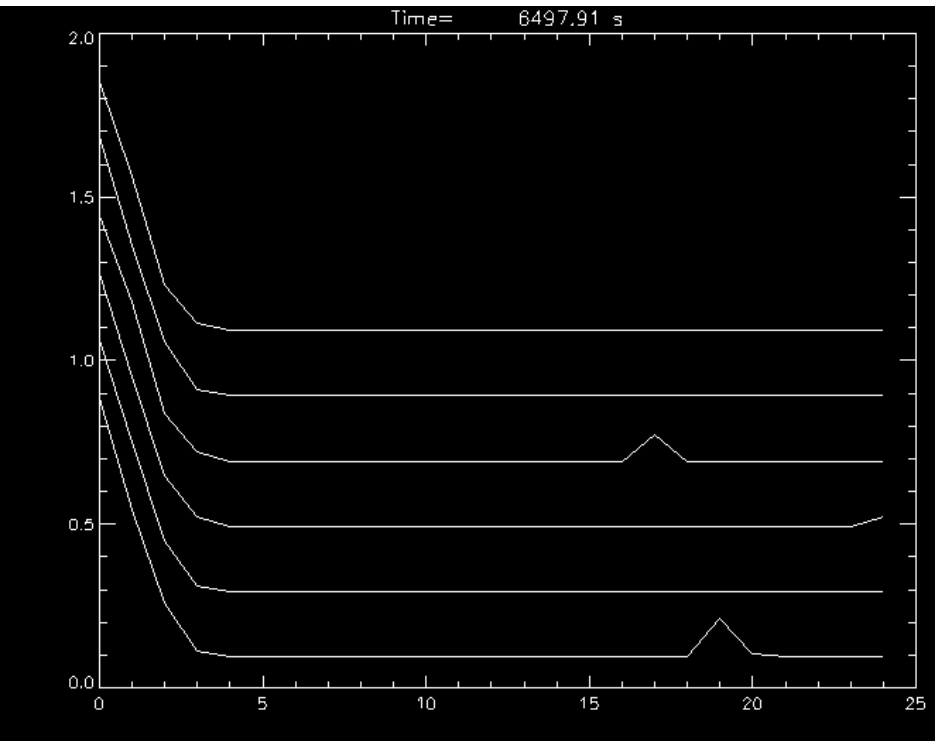
Planetary
and
Space Science

www.elsevier.com/locate/pss

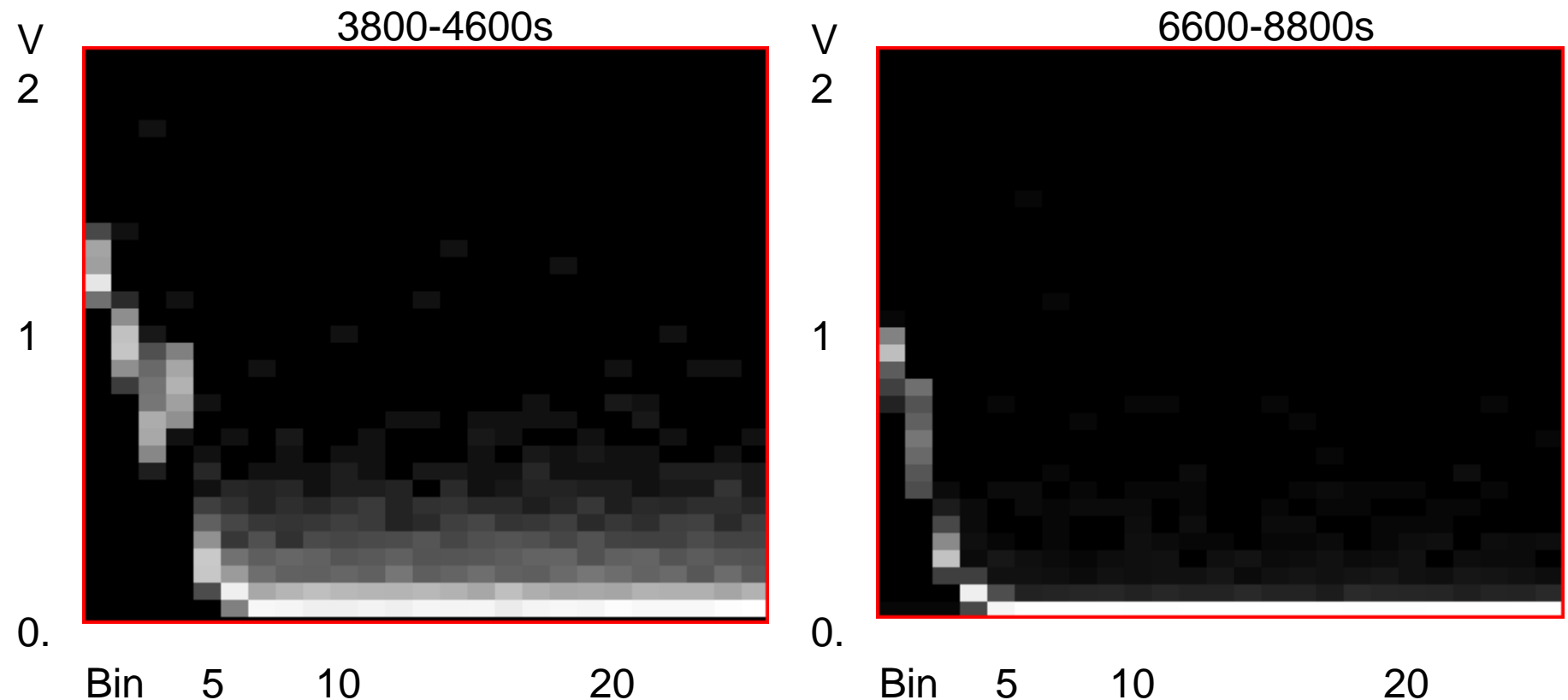
Descent motions of the Huygens probe as measured by the Surface Science Package (SSP): Turbulent evidence for a cloud layer

Ralph D. Lorenz^{a,b,*}, John C. Zarnecki^b, Martin C. Towner^b, Mark R. Leese^b, Andrew J. Ball^b, Brijen Hathi^b, Axel Hagermann^b, Nadeem A.L. Ghafoor^c

Not obviously voltage noise - likely to be real movement/stress on transducer. But is it echo from cloud, echo from turbulent eddies, or mechanical noise on the sensor ?



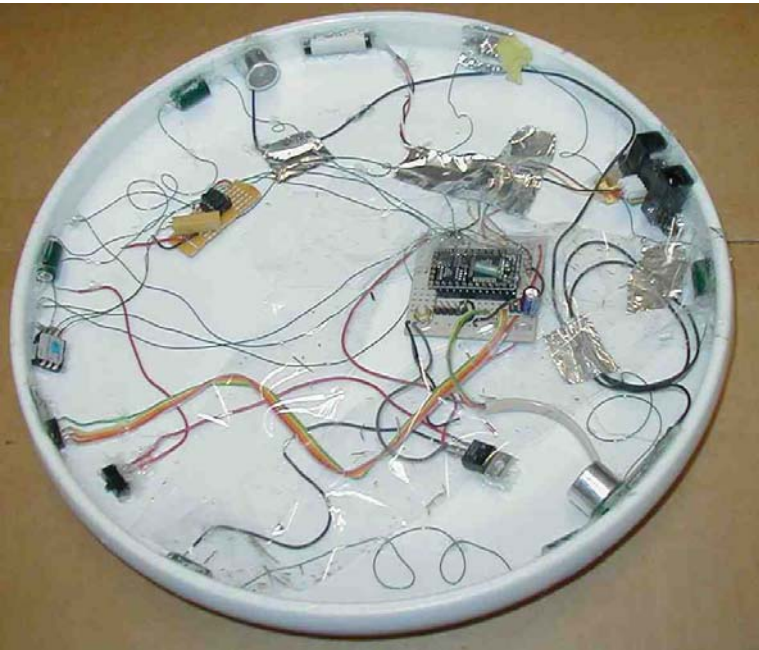
Histogram of signal voltage vs time bin



Signals as likely to occur in distant time bins as they are in nearby bins.
Appears to reject the hypothesis that most signals are echoes.

A real environmental signal (but peak in acoustic data cut off by noise before peak in Tilt signatures reached) but not echoes - structural creaking excited by turbulence?

Use microphone as indicator of flow separation ?



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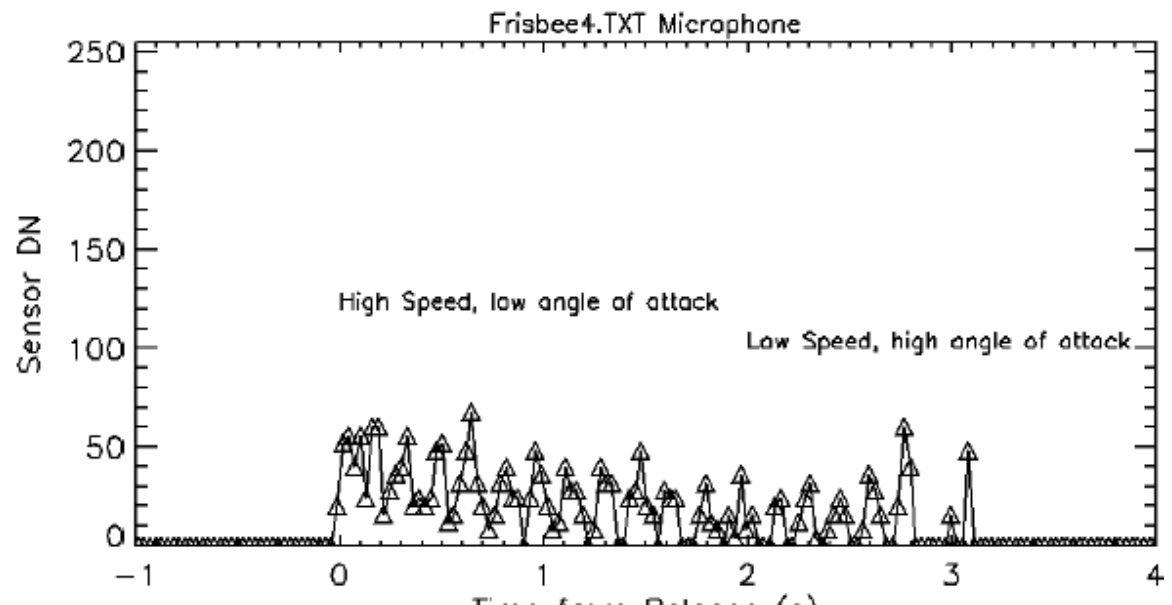
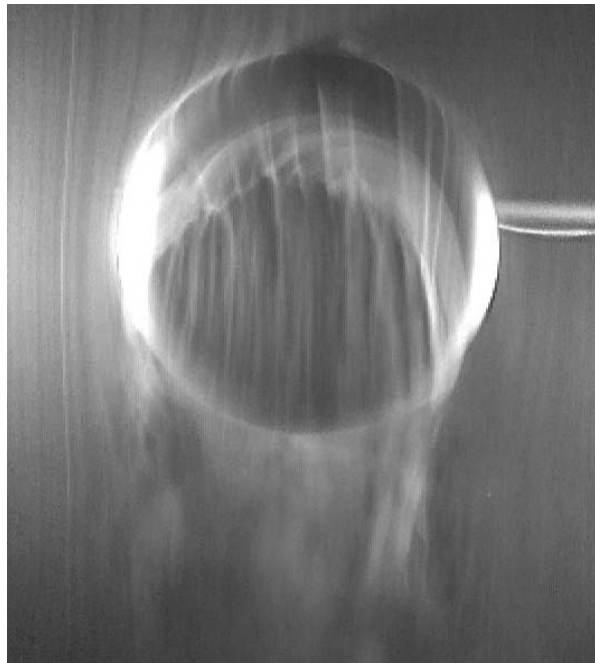
MEASUREMENT SCIENCE AND TECHNOLOGY

Meas. Sci. Technol. 16 (2005) 738–748

[doi:10.1088/0957-0233/16/3/017](https://doi.org/10.1088/0957-0233/16/3/017)

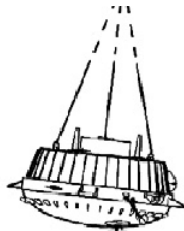
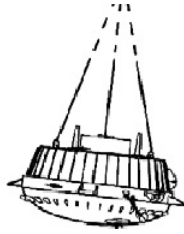
Flight and attitude dynamics measurements of an instrumented Frisbee

Ralph D Lorenz



Acoustical Measurements are challenging from a rapidly moving vehicle.

- Perception of a parachute-borne probe as a gently descending object is wrong. Fast and noisy
- Balloon-drop or similar testing should have been carried out
- ACP pump was identified as a potential noise source during development - why was it a problem for SSP? (and why is signature not obvious on HASI - structurally conducted rather than acoustically radiated ?)
- Science impact of HASI ACU has been modest, but outreach value not insignificant
- Zero offset - important in measuring low surface winds



Being sure about a few detections is better than having many possibles.

Want multiple echoes (>3) of same reflecting structure (i.e. high PRF; though NB range ambiguity is a challenge for 'relativistic' regime ($V \sim c$))

Adapt sampling strategy to descent rate
- burst mode

V / PRF

Acoustic measurements too 'novel', too nonspecific to merit dedicated acoustic instrument ?

Need to piggyback microphone on 'package' instruments like SSP or atmospheric structure, and/or invoke its outreach appeal.

(NB speed of sound a useful measure of methane humidity on Titan; also diagnostic of ortho:para hydrogen ratio in giant planet atmospheres. Acoustic anemometers are also a promising technique)

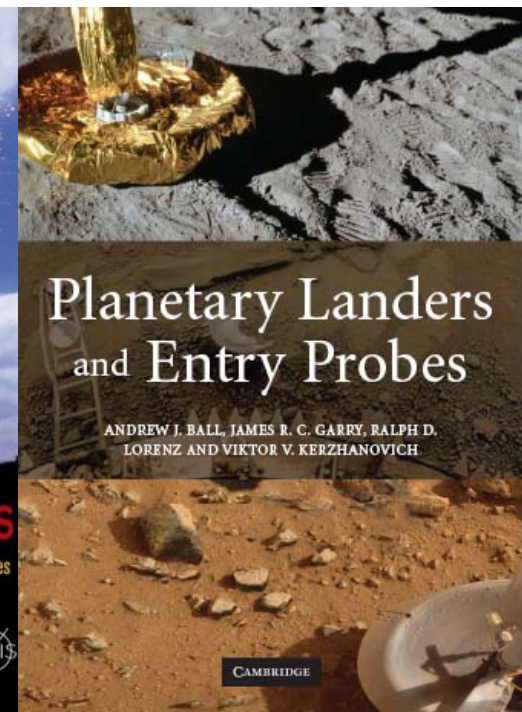
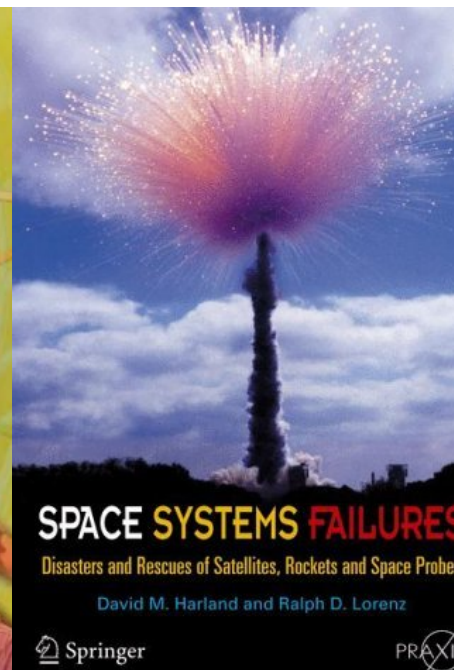
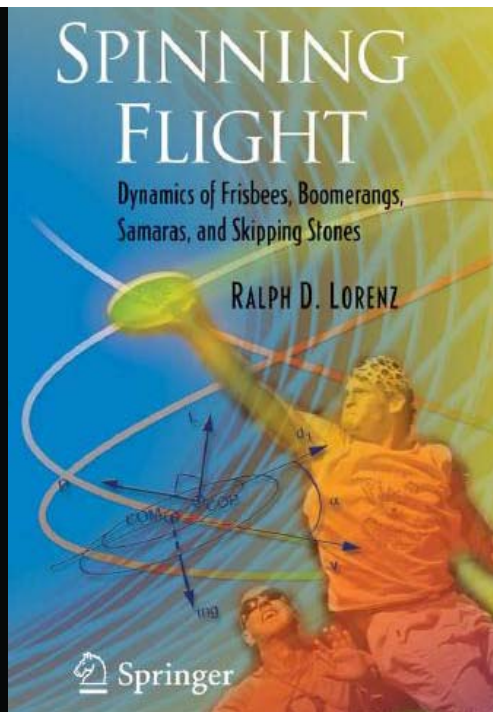
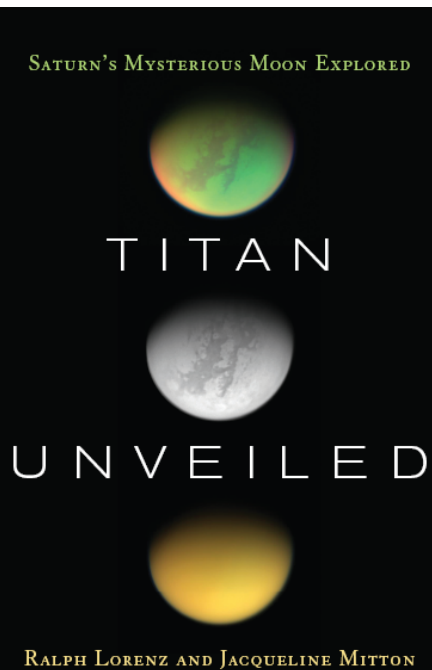


Sodar, methane humidity
and anemometer on a
Titan Montgolfiere ?

Acoustic instrumentation on a Titan lander - methane humidity, anemometry, detection of saltation or waves (dunes v liquid) Infrasound from bolides, cracks from tidal stresses. Booming dunes? Also engineering/outreach of mechanism operation (doors, pumps, sampling arm, Titan Bumblebee UAV takeoff..)



Fin





Tucson, September 2007

Linear array of 20 stations deployed
E-W (PICAXE 18X datalogger)

Set to record 1 hour of 1/s 1-byte
data of pressure, temperature, light,
microphone (numerically
differenced-summed)

Encountered large dust devil after
~30 minutes

